CHAPTER 52

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CHAPTER 52 - DOORS

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GENERAL - DESCRIPTION AND OPERATION

The Duke is equipped with a fail safe cabin door latching mechanism. When the door latch bolts are in position, a spring-loaded secondary locking device maintains a safety locked condition. In addition, a pressure slide lock prevents inadvertent movement of either the secondary system or the door handle itself when pressurized. When the door is closed, the outside cabin door handle is spring loaded to fit into a recess in the door to create a flat, aerodynamically clean surface. The door may be locked with a key.

To open the door from the outside, press inward on the forward end of the handle to raise the aft end enough to grasp it. On serials P-123, P-127 and after, push the safety release button and lift the handle from its recess and turn it counterclockwise until the door opens. The door will swing out and forward over the left wing section. The door may be closed from the outside by rotating the handle clockwise. The three door latching bolts activate three switches mounted on the bulkhead behind the fuselage door frame. A fourth switch mounted on the door (serials P-4 through P-126, except P-123) is activated by the door handle latch mechanism. A cabin door warning light on the annunciator panel illuminates when the cabin door is not secure. All door switches must be activated to turn off the annunciator light.

To close the door from the inside, pull the door shut firmly with the handle in the forward position. Rotate the door handle aft in a counterclockwise manner until the safety lock bolt handle moves aft or the safety lock button pops outward. When the door handle has been rotated completely aft, (serials P-4 through P-126, except P-123) the safety lock bolt handle will snap forward to its original position.

At this point, the door is securely locked and cannot be

opened except by moving the safety lock bolt full aft; or o the serials P-123, P-127 and after, pressing the safety loc button in. If there is residual pressure remaining in th cabin, the red "T" handle, located forward of the cabi door handle, must be pulled to override the pressur locking mechanism before the safety lock bolt or safet lock button will move. Once the safety lock bolt has bee pulled aft, or the safety lock button pressed in, the doo handle may be rotated forward to open the door.

CAUTION

If the cabin door handle is rotated in an attempt to open the door and the safety lock bolt (P-4 through P-126, except P-123) is not in the full aft position, damage may result to the safety lock bolt mechanism.

The Duke 60 Series aircraft are equipped with a retractable assist step, attached to the fuselage under the cabin door, to aid in entering and leaving the aircraft. A cable, attached to the actuator arm on the right hand main landing gear strut extends the assist step when the landing gear is extended When the landing gear is retracted, the assist step is retracted and fits flush with the fuselage.

The CABIN DOOR light in the annunciator remains illuminated until the door is closed, latched and locked, since all three latch pin switches are wired in parallel with one another; and on serials P-4 through P-126, except P-123, with the door locked switch. When the cabin door is closed and latched, each latch pin compresses the actuator on its respective switch mounted on the bulkhead behind the aft frame of the doorway. When the cabin door handle is rotated to the locked position (serials P-4 through P-126, except P-123) a spacer at the bottom of the latch mechanism lock bolt compresses the arm on the door locked switch.

ENTRANCE STAIRS - MAINTENANCE PRACTICES

RETRACTABLE ASSIST STEP REMOVAL (Figure` 201)

- a. Remove the bolt (1) from the outboard side of the step (2).
- Slide the step off far enough to clear the stops (6).
 Allow the step to rotate until the tension on the spring (5) is relieved.
 - c. Remove the step.

RETRACTABLE ASSIST STEP INSTALLATION

- a. Place the spring in the step. Align the spring end in the corresponding hole in the step retainer (3).
- b. Slide the step on the strut assembly shaft (4) and rotate counterclockwise until the remaining spring end is in

- place in the hole in the strut assembly shaft.
- c. Continue rotating the step until the stops an aligned. Slide the step on until the stops engage.
 - d. Replace the bolt in the outboard side of the step.

STEP ADJUSTMENT (FOLDED POSITION)

If the step is not flush with the skin when in the folder position it may be adjusted. This is done by loosening the two bolts in the stop (24), and sliding it up or down as needed.

STRUT ASSEMBLY REMOVAL (Figure 201)

- a. Remove the small access plate below the cabin door in the area of the strut assembly (7).
- 1. Step Retaining Bolt Strut Assembly Extension 2. Cabin Step Spring Step Retainer Clevis Adjusting Rod End 17. Step Assembly Shaft 18. Cable Clevis Step Extension Spring 19. **Bell Crank Actuator Cable** Step Extension Stops 20. U-bolt Cable Clamps Strut Assembly Covered Cable 21 8 Bell Crank Strut Assembly Screw 22. Cable Actuator Arm Lower Bell Crank Arm. 23. Steel Ball Stop (Cable) Strut Assembly Retaining Pin 10. 24. Folding Step Stop Cotter Pin 25. Bell Crank Nut 29 11. Strut Assembly Retaining Pin 26. Bell Crank Bolt 23 Washer 27. Bell Crank Washer 12. Strut Assembly Retaining Pin **Bell Crank Support** Strut Assembly Bearing -1<u>1.</u> -10 Assembly Housing & Lubrication Plug Bell Crank Pulley Shaft 29. Cable Retaining Clip Retainer Pin 15. Bell Crank Pulley 30. Bell Crank Pulley Shaft Upper Bell Crank Arm 26 *P-4 thru P-509 *P-510 and after

Retractable Assist Step Figure 201

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- b. Remove the access plate from the bottom of the airplane in the area of the strut assembly.
- c. Remove the assembly screw (8) which connects the lower bell crank arm (9) to the strut assembly.
- d. Remove the cotter pin (10), washer (11) and pin (12) which secures the strut assembly to the bearing housing (13).
- e. Remove the six strut assembly attaching bolts revealed by removing the bottom panel, and remove the bearing housing.

STRUT ASSEMBLY INSTALLATION

- a. Position the bearing housing (13) through the bottom panel and install the six strut assembly attaching boits. Install the bottom panel.
- b. Position the strut assembly (7) to the bearing housing and secure with the pin (12), washer (11), and cotter pin (10).
- c. Position the bell crank arm (9) to the strut assembly and secure with the attaching screw (8).
- d. Install the access plate to the bottom of the airplane in the area of the strut assembly.
- e. Install the small access plate below the cabin door in the area of the strut assembly.

CABLE REMOVAL (P-4 THRU P-509) (Figure 201)

- a. Remove the large access plate below the cabin door
- b. Remove the L-shaped cable retaining clip (14) from the bell crank pulley (15).
 - c. Lower the flaps.
- d. Release tension from the extension spring (16) by backing off the clevis adjusting rod end (17) from the clevis.
- e. Separate the bell crank actuator cable at the cable clevis (18).
- f. Disconnect the bell crank actuator cable (19) from the extension spring.

NOTE

The cable/return spring connection is accessible through holes which are exposed when the flaps are in the down position.

- g. Remove the bell crank actuator cable.
- Remove the long access plate inboard and slightly aft of the RH main landing gear door.
- i. Remove the U-bolt clamps (20) from each end of the covered cable (21).

- j. Detach the covered cable from the actuator arm
 (22) on the landing gear strut.
 - k. Note the routing of the covered cable and remove.

CABLE INSTALLATION (P-4 THRU P-509)

a. When installing either of the cables, peel laminations may be removed as required to allow the cable actuator arm (22) to rotate with 3 ± 2 inch-pounds torque.

NOTE

When installing the cables, be certain the steel ball stop (23) on the cable is in place on the bell crank pulley and tighten the rod end into the clevis to a depth of one inch. This is all that is necessary to rig the step travel.

- b. Position the covered cable and route as noted during removal.
- Attach the covered cable to the actuator arm (22)
 on the landing gear strut.
- d. Install the U-bolt clamps (20) to each end of the covered cable.
- e. Install the long access plate inboard and slightly aft of the RH main landing gear door.
- f. Position the bell crank actuator cable and connect to the extension spring (16) and the clevis adjusting rod end (17).
 - g. Raise the flaps.
- h. Install the L-shaped cable retaining clip (14) to the bell crank pulley (15).
- i. Install the large access plate below the cabin door.

CABLE REMOVAL (P-510 AND AFTER) (Figure 201)

- a. Remove the large access plate below the cabin door.
- b. Remove the L-shaped cable retaining clip (14) from the bell crank pulley (15).
- c. Remove the long access plate inboard and slightly aft of the RH main landing gear door.
- d. Release the tension from the extension spring (16) by backing off the clevis adjusting rod end (17) from the cable actuator arm (22), located on the landing gear strut.
 - e. Lower the flaps.
- f. Separate the bell crank actuator cable at the cable clevis (18).
- g. Disconnect the bell crank actuator cable (19) from the extension spring.

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NOTE

The cable/return spring connection is accessible through holes which are exposed when the flaps are in the down position.

- h. Remove the bell crank actuator cable.
- i. Remove the U-bolt clamps (20) from each end of the covered cable (21).
- j. Detach the covered cable from the actuator arm
 (22) on the landing gear strut.
 - k. Note the routing of the covered cable and remove.

CABLE INSTALLATION (P-510 AND AFTER)

a. When installing either of the cables, peel laminations may be removed as required to allow the cable actuator arm (22) to rotate with 3 ± 2 inch-pounds torque.

NOTE

When installing the cables, be certain the steel ball stop (23) on the cable is in place on the bell crank pulley and tighten the rod end into the clevis to a depth of one inch. This is all that is necessary to rig the step travel.

- b. Position the covered cable and route as noted during removal.
- c. Position the bell crank actuator cable (19) and connect to the extension spring (16) and the non-adjustable clevis end of the covered cable.
- d. Attach the clevis adjusting rod end (17) of the covered cable to the actuator arm (22) on the landing gear strut.
- e. Install the U-bolt clamps (20) to each end of the covered cable.
- f. Install the long access plate inboard and slightly aft of the RH main landing gear door.

- g. Raise the flaps.
- h. Install the L-shaped cable retaining clip (14) to the bell crank pulley (15).
- i. Install the large access plate below the cabin door.

BELL CRANK ASSEMBLY REMOVAL (Figure 201)

- a. Remove the screw that attaches the bell crank to the strut assembly.
- b. Remove the bell crank nut (25), bolt (26) and washer (27) from the upper bell crank arm (31). Remove the lower bellcrank arm (9).
- Remove the access plate on the under side of the airplane, near the bell crank.
- d. Remove the access plate below the cabin door in the area of the bell crank.
- e. Remove the bell crank actuator cable as described in CABLE REMOVAL.
- f. Locate the bell crank support assembly (28) in the lower access opening and remove the bell crank pulley shaft retainer pin (29) from the bell crank pulley shaft (30).
- g. Remove the upper bell crank arm (31), the bell crank pulley (15) and bell crank pulley shaft (30).

BELL CRANK ASSEMBLY INSTALLATION

- a. Position the upper bell crank arm (31), pulley (15) and shaft (30) to the bell crank support assembly (28) and install the retainer pin (29) to the bell crank shaft (30).
- b. Install the bell crank actuator cable as described in CABLE INSTALLATION.
- c. Install the access plate below the cabin door in the area of the bell crank.
- d. Install the access plate on the underneath side of the airplane, near the bell crank.
- e. Install the lower bell crank arm (9) to the upper bell crank arm (31) and secure with the attaching bolt (26), washer (27) and nut (25).
- f. Install the screw that attaches the bell crank to the strut assembly.

DOOR WARNING - MAINTENANCE PRACTICES

The door locked and door latched switches will not normally require adjustment except when a new switch is installed.

a. Adjust the latch pin switches as follows:

NOTE

Before making adjustments to the latch pin switches be sure the latch pins are properly adjusted as outlined under DOOR LATCH ADJUSTMENT in this chapter.

- Remove the cabin upholstery panels adjacent to the upper, lower and center latches to gain access to the switches.
 - 2. Close and lock the cabin door.
- Rotate the cabin door handle clockwise against the lock mechanism stop while in the locked position. This will eliminate the effect of play in the lock mechanism.
- Back the switch adjustment off until the switch is not actuated.
 - 5. Readjust the switch until it actuates.
- 6. Adjust upper and lower switch overtravel to .12 to .20 inch and middle switch overtravel to .07 to .11 inch.

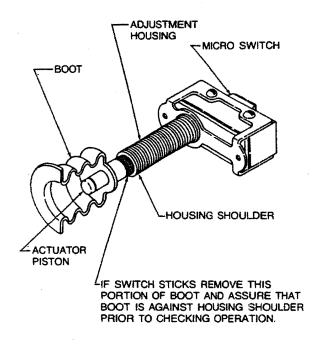
NOTE

One complete revolution of the adjusting nut will provide .031 inch travel.

7. Reopen the cabin door and note if the switch remains in the actuated position due to the friction of the rubber boot on the actuating piston. If this occurs remove the safety wire on the boot, trim one-half of that portion of the boot which rides the actuating piston as indicated in

Figure 201 and slide the boot along the piston until it rests against the shoulder of the adjustment housing.

- Install new safety wire on the boot and close, lock and open the door to check for proper operation.
 - 9. Install the cabin upholstery panels.



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Switch Boot Modification Figure 201

- b. Adjust the door locked switch (P-4 thru P-126, except P-123) as follows:
- Remove the upholstery panel under the window of the cabin door to gain access to the door locked switch.
- Loosen the attaching screws and position the switch in its mounting slots so that the CABIN DOOR light goes out when the door is closed, latched, and locked.